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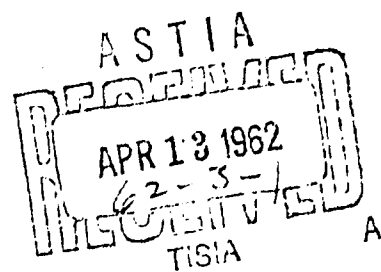
STRESSES IN A LINEAR INCOMPRESSIBLE VISCO-ELASTIC
CYLINDER WITH ANNIHILATING INNER SURFACE

by

M. Shinozuka

Office of Naval Research
Project Nonr 064-448
Contract Nonr 266(78)
Technical Report No. 12
CU-12-62-ONR 266(78) CE

March 1962



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SUMMARY

A method is developed to find the stresses and strains in an incompressible visco-elastic hollow cylinder with annihilating inner radius contained by an elastic case and subject to internal pressure under the assumption of a state of plane strain.

Stresses and strains are computed for a material with deviatoric stress-strain relations characteristic of a standard solid. The numerical computation is carried out with the aid of an I.B.M. digital computer 1620 and is intended to illustrate the effects of the thickness of the cylinder, of the rate of increase of the internal pressure and of the strength of the reinforcement provided by the elastic shell.

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1. Introduction

The problem of evaluation of stresses and strains in a solid propellant with a burning inner surface under internal pressure has not yet been solved in a general way because of the evident difficulty of the treatment of the boundary condition at the moving inner surface.

It is the purpose of the present investigation to find a solution of this problem by evaluating stresses and strains in an elastically case-bonded linear visco-elastic hollow cylinder with annihilating inner surface subject to internal pressure.

For simplicity, incompressibility of the material and a state of plain strain are assumed; hence

$$\begin{aligned} \epsilon_r &= e_r, \quad \epsilon_\theta = e_\theta, \quad \epsilon_z = e_z = 0 \\ \epsilon_r &= du/dr, \quad \epsilon_\theta = u/r \\ \epsilon_r + \epsilon_\theta &= \frac{du}{dr} + \frac{u}{r} = 0 \end{aligned} \quad (1)$$

where ϵ_r , ϵ_θ , e_r and e_θ are respectively the radial and tangential components of strain and deviatoric strain, and u is the radial displacement.

Eq. (1) is satisfied by $u = k(t)/r$ where $k(t)$ is a function of time t only. Hence,

$$\epsilon_r = -k(t)/r^2, \quad \epsilon_\theta = k(t)/r^2 \quad (2)$$

When s_{ij} and e_{ij} respectively denote the components of deviatoric stress and strain, the stress-strain relations for incompressible linear visco-elastic materials are of the well-known form

$$P(s_{ij}) = Q(e_{ij}) \quad (3)$$

where \bar{P} and \bar{Q} are linear differential operators with respect to time t or, in terms of Laplace transforms with initially zero condition,

$$\bar{P}(p) \bar{\epsilon}_{ij}(p) = \bar{Q}(p) \bar{\epsilon}_{ij}(p) \quad (3')$$

where p denotes the transform parameter.

The equation of equilibrium

$$\frac{d\sigma_r}{dr} + \frac{\sigma_r - \sigma_\theta}{r} = 0 \quad \text{or} \quad \frac{d\sigma_r}{dr} + \frac{\sigma_r - \sigma_\theta}{r} = 0 \quad (4)$$

when \bar{P} operates on both sides with assumed interchangeability between differentiations with respect to time t and radius r becomes

$$\frac{d}{dr} \{ \bar{P}(\sigma_r) \} = \{ \bar{P}(\sigma_\theta) - \bar{P}(\sigma_r) \} / r = \{ \bar{Q}(\epsilon_\theta) - \bar{Q}(\epsilon_r) \} / r$$

after the stress-strain relation Eq. (3) has been introduced.

Substituting Eq. (2) into the above equation gives

$$\frac{d}{dr} \{ \bar{P}(\sigma_r) \} = \frac{2}{r^3} \bar{Q} \{ k(t) \}$$

from which $\bar{P}(\sigma_r)$ is obtained by integration with respect to r :

$$\bar{P}(\sigma_r) = -\frac{1}{r^3} \bar{Q} \{ k(t) \} + C(t) \quad (5)$$

where $C(t)$ is a function of t only.

The equilibrium of the elastic shell requires the relation

$$\sigma_r]_{r=b} = -\frac{h}{b} \sigma_c$$

between the shell stress σ_c and the radial stress of the cylinder $\sigma_r]_{r=b}$ at the interface between the cylinder and the shell, while σ_c is in turn

related to the strain of the cylinder $\epsilon_\theta]_{r=b}$ at this interface by the relation

$$\sigma_c = \frac{E_c}{1 - \nu_c^2} \epsilon_\theta]_{r=b}$$

since $\epsilon_\theta]_{r=b}$ is identical to the shell strain ϵ_c ; h is the thickness of the shell, b the outer radius of the hollow cylinder and E_c and ν_c are Young's modulus and Poisson ratio of the shell respectively.

Therefore the first boundary condition at $r = b$ has the form

$$\sigma_r]_{r=b} = -E_c' \epsilon_\theta]_{r=b} = -E_c' \frac{k(t)}{b^2} \quad (6)$$

or

$$P(\sigma_r)]_{r=b} = -\frac{E_c'}{b^2} P\{k(t)\} \quad (6')$$

where the second of Eqs. (2) has been introduced for ϵ_θ , and

$$E_c' = \frac{E_c}{1 - \nu_c^2} \cdot \frac{h}{b} \quad (7)$$

$C(t)$ is obtained from Eqs. (5) and (6'):

$$C(t) = \frac{1}{b^2} Q\{k(t)\} - \frac{E_c'}{b^2} P\{k(t)\}$$

Hence

$$P(\sigma_r) = \left(\frac{1}{b^2} - \frac{1}{r^2}\right) Q\{k(t)\} - \frac{E_c'}{b^2} P\{k(t)\}. \quad (8)$$

However, as pointed out by E. H. Lee, R. M. Radok and W. B. Woodward [1], it is not possible to operate with P on the second boundary condition at

the inner surface

$$-\sigma_r]_{r=a(t)} = p(t) \quad (9)$$

and substitute it into Eq. (5) as in the first boundary condition, since Eq. (5) is essentially a relation between stress and strain associated with a fixed material particle, while at each instant $\sigma_r]_{r=a(t)}$ in Eq. (9) represents the radial stress of a different material particle; $a(t)$ is a monotonically increasing function of time representing the inner radius of the hollow cylinder until

$$a(t_0) = b \quad (10)$$

is reached when the annihilation is completed at $t = t_0$.

2. General Expressions for Stresses and Strains

The difficulty is removed however if the stress-strain relation is expressed in the form

$$s_{ij} = \mathcal{L}(e_{ij}) \quad (11)$$

instead of Eq. (3), where \mathcal{L} is a linear operator in derivatives and integrals with respect to time only, or in terms of its Laplace transform under zero initial conditions:

$$\bar{s}_{ij}(p) = \bar{\mathcal{L}}(p) \bar{e}_{ij}(p) \quad (11')$$

The substitution of Eq. (2) into Eq. (11) gives

$$s_r = -\frac{1}{r^2} \mathcal{L}\{k(t)\}, \quad s_\theta = \frac{1}{r^2} \mathcal{L}\{k(t)\}$$

The radial stress σ_r is then obtained from Eq. (4):

$$\sigma_r = -\frac{1}{r^2} \mathcal{L}\{k(t)\} + D(t)$$

where $D(t)$ is a function of t only.

Use of the first boundary condition Eq. (6) determines $D(t)$:

$$D(t) = \frac{1}{b^2} \mathcal{L}\{k(t)\} - E_c \frac{k(t)}{b^2}$$

Hence

$$\sigma_r = \left(\frac{1}{b^2} - \frac{1}{r^2}\right) \mathcal{L}\{k(t)\} - E_c \frac{k(t)}{b^2} \quad (12)$$

The unknown function $k(t)$ is now obtained from the integro-differential equation

$$-p(t) = \left(\frac{1}{b^2} - \frac{1}{a^2(t)}\right) \mathcal{L}\{k(t)\} - \frac{E_c}{b^2} k(t) \quad (13)$$

which is the result of the application of the second boundary condition Eq. (9) to Eq. (12).

E. H. Lee, R. M. Radok and W. B. Woodward [1] solved a special case of the problem in which the mechanical model of the material was assumed to be a Kelvin body. This implies a stress-strain relation of the form Eq. (11) with

$$\mathcal{L} = 2G + 2\eta \frac{\partial}{\partial t}$$

or of the form of Eq. (11') with

$$\mathcal{L}(p) = 2G + 2\eta p;$$

G denotes the shear modulus and η the coefficient of viscosity in shear as shown in Fig. 1.

In more generality linear visco-elastic material can be represented [2]

by n Maxwell elements coupled in parallel with discrete relaxation times

$\tau_i = \eta_i / G_i$ (Fig. 2) and normalized discrete relaxation spectrum $F_i(\tau_i) = G_i(\tau_i) / G$

(Fig. 3) or by the limit of the same model when n approaches infinity with

continuous relaxation time τ and normalized continuous relaxation spectrum

$f(\tau) = g(\tau) / G$ where the unrelaxed shear modulus $G = \sum_{i=1}^n G_i$ or $G = \int_0^{\infty} g(\tau) d\tau$.

It can then be shown [3] that the stress deviation is related to the strain

deviation in the form

$$s_{ij} = 2G \left[e_{ij} + \int_0^t \dot{\psi}(t-\theta) e(\theta) d\theta \right] \quad (14)$$

or in the form of Eq. (11') with

$$\bar{\mathcal{L}}(\rho) = 2G \{ 1 + \bar{\psi}(\rho) \}.$$

The relaxation-rate-function $\dot{\psi}(t) = d\psi(t)/dt$ (this notation for the time

derivative is used henceforth) in Eq. (14) is the time derivative of the re-

laxation-function $\psi(t)$ which is defined so as to produce the stress response

to the unit step strain input $e_{ij}(t) = e_0 = \text{const.}$ in the form $s_{ij}(t) = 2Ge_0\psi(t)$.

$\psi(t)$ can also be obtained from a knowledge of the relaxation spectrum:

$$\dot{\psi}(t) = - \sum_{i=1}^n \frac{F_i}{\tau_i} e^{-t/\tau_i} \quad (15)$$

for the discrete relaxation spectrum and

$$\dot{\psi}(t) = - \int_0^{\infty} \frac{f(\tau)}{\tau} e^{-t/\tau} d\tau \quad (16)$$

for the continuous relaxation spectrum. Eq. (12) can now be written in the

form

$$s_r = 2G \left(\frac{1}{\nu} - \frac{1}{r} \right) \left[k(t) + \int_0^t \dot{\psi}(t-\theta) k(\theta) d\theta \right] - \frac{E_r}{b} k(t) \quad (12')$$

The determining equation Eq. (13) for $k(t)$ then becomes

$$-p(t) = 2G\left(\frac{1}{b^2} - \frac{1}{a^2(t)}\right)\left[k(t) + \int_0^t \dot{\gamma}(t-\theta)k(\theta)d\theta\right] - \frac{E_c''}{b^2}k(t) \quad (13')$$

which can be reduced to the form of Volterra's integral equation:

$$k(t) + \frac{\frac{1}{a^2(t)} - \frac{1}{b^2}}{\frac{1}{a^2(t)} + \frac{\mu-1}{b^2}} \int_0^t \dot{\gamma}(t-\theta)k(\theta)d\theta = \frac{p(t)}{2G\left(\frac{1}{a^2(t)} + \frac{\mu-1}{b^2}\right)} \quad (17)$$

where $\mu = E_c''/2G$.

The solution $k(t)$ of Eq. (17) furnishes the strains with the aid of Eq. (2). The radial stress σ_r is obtained from Eq. (12'), which involves the integration of $k(t)$. However, the identity from Eq. (13')

$$2G\left[k(t) + \int_0^t \dot{\gamma}(t-\theta)k(\theta)d\theta\right] = \left\{\frac{E_c''}{b^2}k(t) - p(t)\right\} / \left(\frac{1}{b^2} - \frac{1}{a^2(t)}\right) \quad (18)$$

can be used to avoid the evaluation of this integral writing the radial stress σ_r in the form

$$\sigma_r = \frac{\frac{1}{b^2} - \frac{1}{r^2}}{\frac{1}{b^2} - \frac{1}{a^2(t)}} \left[\frac{E_c''}{b^2}k(t) - p(t) \right] - \frac{E_c''}{b^2}k(t) \quad (t \neq t_0) \quad (19)$$

The tangential stress σ_θ is obtained from the equation of equilibrium Eq. (4):

$$\sigma_\theta = \frac{\frac{1}{b^2} + \frac{1}{r^2}}{\frac{1}{b^2} - \frac{1}{a^2(t)}} \left[\frac{E_c''}{b^2}k(t) - p(t) \right] - E_c'' \frac{k(t)}{b^2} \quad (t \neq t_0). \quad (20)$$

when t approaches t_0 , the right hand side of Eq. (18) should be interpreted as the limit:

$$q(t_0) = 2G[k(t_0) + \int_0^{t_0} \psi(t_0 - \theta)k(\theta)d\theta] = \lim_{t \rightarrow t_0} \frac{E_c \frac{k(t)}{b^2} - p(t)}{\frac{1}{b^2} - \frac{1}{a^2(t)}} \quad (21)$$

since $\lim_{t \rightarrow t_0} a(t) = b$ from Eq. (10) and $\lim_{t \rightarrow t_0} E_c \frac{k(t)}{b^2} = \lim_{t \rightarrow t_0} p(t)$ from Eq. (13').

Hence, as t approaches t_0 , the stresses are given by

$$\sigma_r = -p(t_0) \quad (22)$$

$$\sigma_\theta = \frac{2}{b^2} q(t_0) - p(t_0) \quad (23)$$

since $\lim_{t \rightarrow t_0} r = b$.

3. Stresses and Strains for Medium with Standard Solid Stress-Strain Relation

When the stress-strain relation in shear is represented by a standard solid (Fig. 4), the relaxation time τ_1 is considered infinite while $\tau_2 = \eta_2/G_2$. The relaxation-rate-function $\dot{\phi}(t)$ is obtained from Eq. (15)

$$\dot{\phi}(t) = -\frac{1-\alpha}{\tau_2} e^{-t/\tau_2} \quad (15')$$

where $\alpha = F_1 = G_1/(G_1 + G_2)$

The substitution of Eq. (15') into Eq. (17) produces

$$h(t) = \frac{1-\alpha}{\tau_2} \cdot \frac{\frac{1}{a^2(t)} - \frac{1}{b^2}}{\frac{1}{a^2(t)} + \frac{1}{b^2}} \int_0^t h(\theta)d\theta = \frac{p(t)e^{t/\tau_2}}{2G \left(\frac{1}{a^2(t)} + \frac{1}{b^2} \right)} \quad (17')$$

where $h(t) = k(t)e^{t/\tau_2}$.

Both sides of Eq. (17') are now divided by the coefficient of the integral on the left hand side and differentiated once with respect to time so as to produce the differential equation

$$f(t)\dot{h}(t) + [\dot{f}(t) - 1] h(t) = \dot{q}(t) \quad (24)$$

where

$$f(t) = \frac{\tau_2}{1-\alpha} \cdot \frac{\frac{1}{a^2(t)} + \frac{1}{b^2} - 1}{\frac{1}{a^2(t)} - \frac{1}{b^2}} \quad (25)$$

and

$$q(t) = \frac{\tau_2}{1-\alpha} \cdot \frac{p(t)e^{t/\tau_2}}{2a \left(\frac{1}{a^2(t)} - \frac{1}{b^2} \right)} \quad (26)$$

With an integrating factor $\exp \left[-\int_0^t \frac{d\tau}{f(\tau)} \right]$, Eq. (24) can be integrated:

$$h(t) = q(t)/f(t) + \frac{e^{-\int_0^t \frac{d\tau}{f(\tau)}}}{f(t)} \int_0^t \frac{q(\tau)}{f(\tau)} e^{\int_0^\tau \frac{d\tau}{f(\tau)}} d\tau$$

and therefore

$$k(t) = q(t)e^{-t/\tau_2}/f(t) + \frac{e^{-t/\tau_2}}{f(t)} e^{\int_0^t \frac{d\tau}{f(\tau)}} \int_0^t \frac{q(\tau)}{f(\tau)} e^{-\int_0^\tau \frac{d\tau}{f(\tau)}} d\tau \quad (27)$$

It is assumed that the annihilating rate of the inner surface is governed by the relation

$$a(t) = a_0 \sqrt{1 - kt} \quad (28)$$

with

$$\kappa = \frac{\rho_0^2 - 1}{\rho_0^2} \cdot \frac{1}{t_0} \quad (29)$$

where a_0 is the initial inner radius (at $t = 0$) and $\rho_0 = b/a_0$. $a(t)$ is plotted against t/t_0 in Fig. 5 with ρ_0 ($= 1.5, 2.0$ and 3) as parameter.

With the form of $a(t)$ given in Eq. (28), the integral $\int_0^t \frac{d\tau}{f(\tau)}$ can be evaluated so that

$$\int_0^t \frac{d\tau}{f(\tau)} = e^{\pm \frac{1-\alpha}{\tau_2} t} \left(1 - \frac{B}{A} t\right)^{\pm \frac{1-\alpha}{\tau_2} \cdot \frac{A-A'}{B}}$$

where

$$A = \frac{1}{a^2(t)} + \frac{\kappa - 1}{b^2}, \quad A' = \frac{1}{a^2(t)} - \frac{1}{b^2} \quad \text{and} \quad B = \frac{\kappa}{a_0^2}.$$

If the internal pressure $p(t)$ is assumed to be of the form

$$p(t) = p_0 (1 - e^{-t/\tau_0}) \quad (30)$$

Eq. (27) becomes

$$h(t) = \frac{p_0 (1 - e^{-t/\tau_0})}{2G(A - Bt)} + \frac{p_0}{2G} \frac{A'}{A^2} \left(1 - \frac{B}{A} t\right) \left(1 - \frac{B}{A} t\right)^{\gamma-1} e^{-\alpha t/\tau_2} \int_0^t \frac{(1 - e^{-\tau/\tau_0}) e^{\alpha \tau/\tau_2}}{\frac{\tau_2}{1-\alpha} \left(1 - \frac{B}{A} \tau\right)^{\gamma+1}} d\tau$$

which can be transformed by introducing the non-dimensional quantities

$\sigma = p_0/(2G)$, $\alpha' = \tau_2/\tau_0$ and $\lambda = t_0/\tau_2$ into

$$\frac{h(t)}{b^2} = \frac{\sigma \gamma (1 - e^{-\alpha' \lambda \frac{t}{t_0}})}{(\rho_0^2 - 1) (1 - \gamma \frac{t}{t_0})} + \frac{\sigma \gamma^2 (1 - \frac{t}{t_0}) (1 - \gamma \frac{t}{t_0})^{\gamma-1}}{\rho_0^2 - 1} e^{-\alpha \lambda \frac{t}{t_0}} y(t) \quad (31)$$

with

$$y(t) = \int_0^{\frac{t}{t_0}} \frac{\lambda (1 - \alpha) (1 - e^{-\alpha' \lambda \tau})}{(1 - \gamma \tau)^{\gamma+1}} e^{\alpha \lambda \tau} d\tau \quad (32)$$

where

$$\gamma = \frac{\lambda(1-\alpha)}{\gamma + \lambda(1-\alpha)} \text{ and } \gamma = \frac{(1-\alpha)(A-A')}{\tau_2 B} = \frac{\lambda(1-\alpha)\mu}{\rho_0^2 - 1}.$$

The integral $\varphi(t)$ can be numerically evaluated. The strains are

$$\epsilon_r = -\frac{\rho_0^2}{R^2} \cdot \frac{k(t)}{b^2}, \quad \epsilon_\theta = \frac{\rho_0^2}{R^2} \cdot \frac{k(t)}{b^2} \quad (33)$$

where $R = r/a_0$, while the stresses

$$\left. \begin{aligned} \sigma_r/\rho_0 \\ \sigma_\theta/\rho_0 \end{aligned} \right\} = -\frac{1 \pm \rho_0^2/R^2}{(\rho_0^2 - 1)(1 - \frac{1}{\rho_0^2})} \left[\frac{\gamma(\rho_0^2 - 1)}{\rho_0^2(1-\alpha)\lambda} \cdot \frac{k(t)}{b^2} - (1 - e^{-\alpha\lambda t}) \right] - \frac{\gamma(\rho_0^2 - 1)}{\rho_0^2(1-\alpha)\lambda} \cdot \frac{k(t)}{b^2} \quad (34)$$

(t ≠ t₀)

$$\left. \begin{aligned} \sigma_r/\rho_0 &= -(1 - e^{-\alpha\lambda}) \\ \sigma_\theta/\rho_0 &= \left(\frac{2}{\rho_0^2} - 1\right)(1 - e^{-\alpha\lambda}) - \frac{2\gamma(1-\gamma)^2}{\rho_0^2 - 1} e^{-\alpha\lambda} \varphi(t_0) \end{aligned} \right\} \quad (t = t_0) \quad (35)$$

Eq. (35) has been obtained considering that by l'Hospital's rule

$$q(t_0) = b^2 \rho_0 \left[\frac{1}{\mu}(1 - e^{-\alpha\lambda}) - \frac{\gamma(1-\gamma)^2}{\rho_0^2 - 1} e^{-\alpha\lambda} \varphi(t_0) \right].$$

4. Numerical Example

The previously defined non-dimensional quantities which characterize the problem are reviewed:

$\rho_0 = b/a_0$ represents the original thickness of the cylinder.

$\lambda = t_0/\tau_2$ shows how much time is required for the total annihilation in terms of the relaxation time of the material.

$\alpha' = \tau_2/\tau_0$ is a measure of the build-up of internal pressure compared to the relaxation time of the material.

$\lambda\alpha' = t_0/\tau_0$ therefore indicates how rapidly the internal pressure is built up to the maximum in comparison with the total annihilation time t_0 .

$\alpha = G_1/(G_1 + G_2)$ is related to the stress relaxation of the standard solid material in such a way that the permanent stress after completion of the stress relaxation under constant strain e_0 is $s_\infty = 2Ge_0\alpha$ while the initial stress is $s_0 = 2Ge_0$.

$\sigma = p_0/2G$ relates the magnitude of the maximum internal pressure to the unrelaxed shear modulus of the material.

$\mu = \frac{E_c''}{2G} = \frac{2G_c}{2G(1 - \nu_c)} \cdot \frac{h}{b}$ indicates the strength of the reinforcement due to the elastic shell where G_c and ν_c are the shear modulus and Poisson ratio of the elastic shell respectively.

The assignment of numerical values for these quantities is listed in Table 1 producing six different cases for which the numerical computations have been carried out using a 1620 I.B.M. digital computer.

Cases I-III are for comparison of the effect of the original thickness of the cylinder. The comparison of Cases I, IV and V shows the effect of either the rate of the application of the internal pressure when t_0 is identical in these three cases (since then τ_0 in Eq. (30) is equal to $t_0/10$, $t_0/5$ and t_0 respectively) or the time t_0 of total annihilation of the cylinder when $\tau_0 = \tau_2$ is identical (since then $t_0 = 10\tau_0$, $5\tau_0$ and τ_0 respectively).

Finally Cases I and VI illustrate the reinforcing action of the elastic shell. In the latter the elastic shell is so strong that it gives rise to negative tangential stresses in the cylinder. Figs. 6(a), (b) and (c) respectively show the effect of the original thickness of the cylinder, the rate of the load application and of the rigidity of the elastic shell on the radial stress σ_r while Figs. 7 and 8 show these effects on the tangential stress and strain respectively. Fig. 9 shows the shell stresses σ_c/p_0 with $h/b = 100$ for the six cases as functions of time t/t_0 . Finally the space distributions of the radial stress for these six cases are also plotted at various times t/t_0 in Fig. 10, while in Figs. 11 and 12 the space distributions of the tangential stress and strain are shown. In Figs. 10-12, vertical straight lines correspond to the positions of the inner surface at the specified time.

TABLE 1

Designation	p_0	λ	μ	
I	1.5	10	.5	$\alpha = .5$
II	2	10	.5	$\alpha' = 1.0$
III	3	10	.5	$\sigma = .1$
IV	1.5	5	.5	for all cases
V	1.5	1	.5	
VI	1.5	10	.4	

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- [1] Lee, E. H., Radok, J. R. M. and Woodward, W. B., "Stress Analysis for Linear Viscoelastic Materials", Transactions of the Society of Rheology, Vol. III, Interscience Publishers, New York, 1959.
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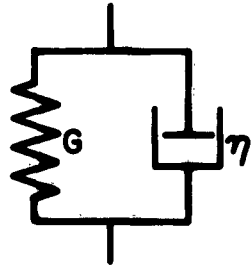


Figure 1. Kelvin Model

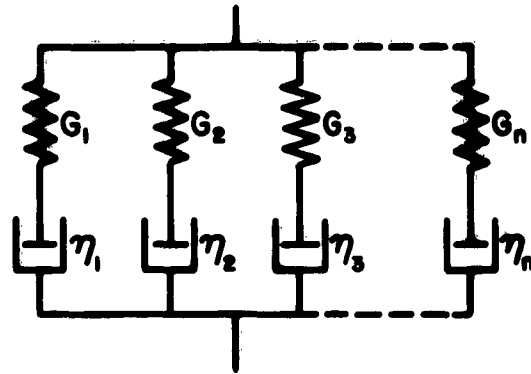


Figure 2. Maxwell Models Coupled in Parallel

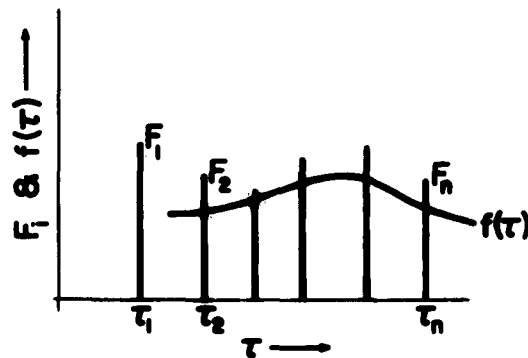


Figure 3. Relaxation Spectrum

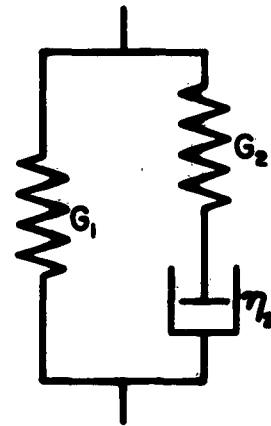


Figure 4. Standard Solid Model

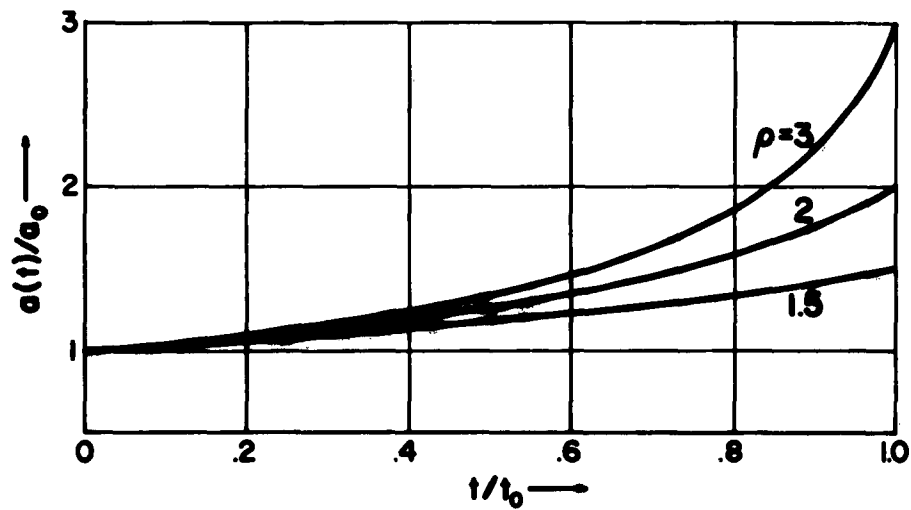
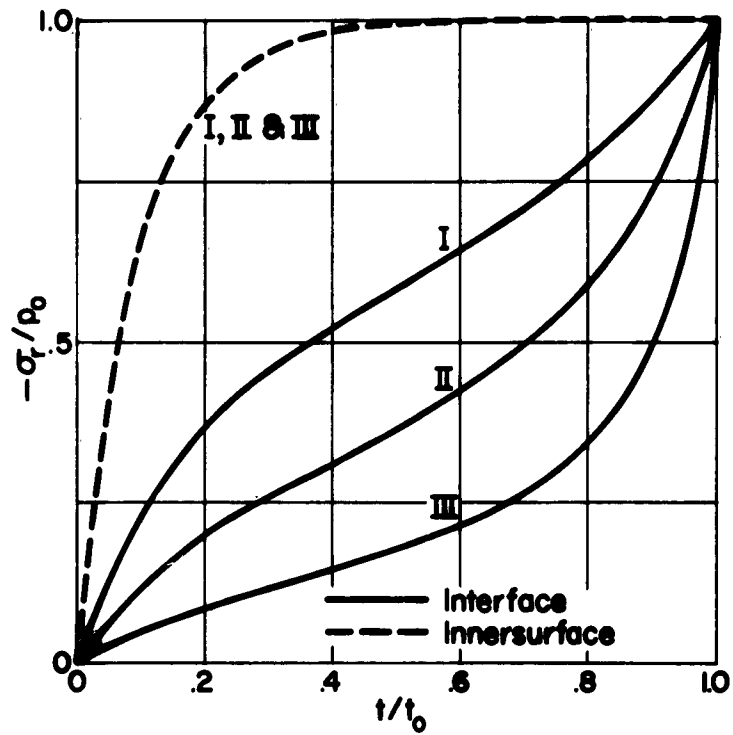
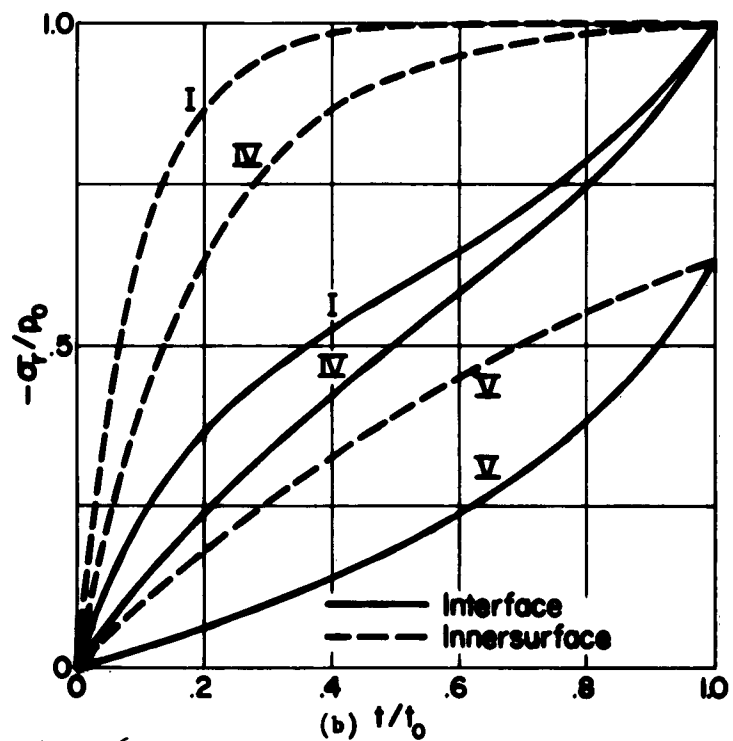


Figure 5. Annihilation of Inner Surface $a(t)/a_0$ as a Function of Time t/t_0



(a)



(b) t/t_0

Figure 6. The Radial Stress σ_r at the Inner Surface and the Interface as a Function of Time t/t_0

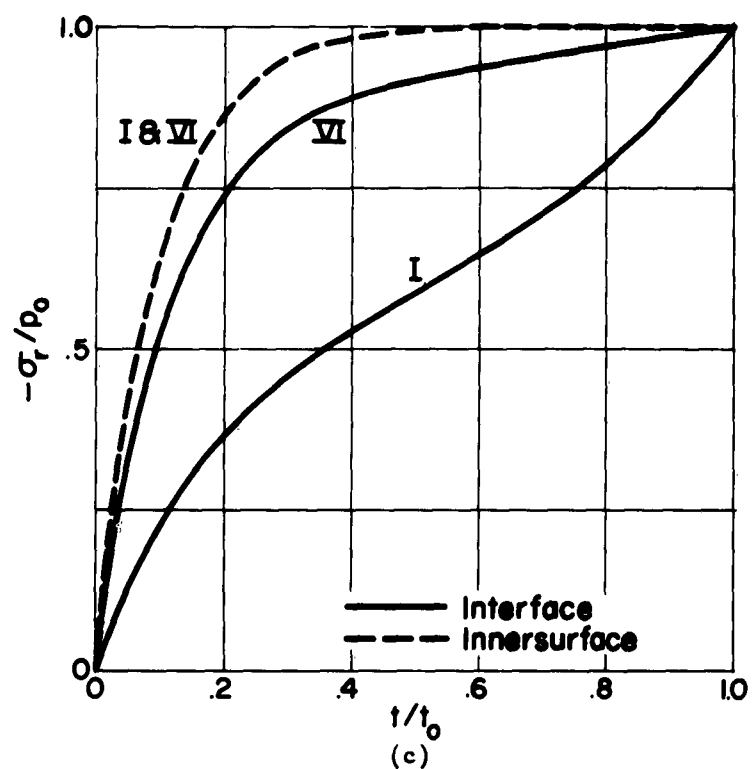


Figure 6. The Radial Stress σ_r at the Inner Surface and the Interface as a Function of Time t/t_0 .

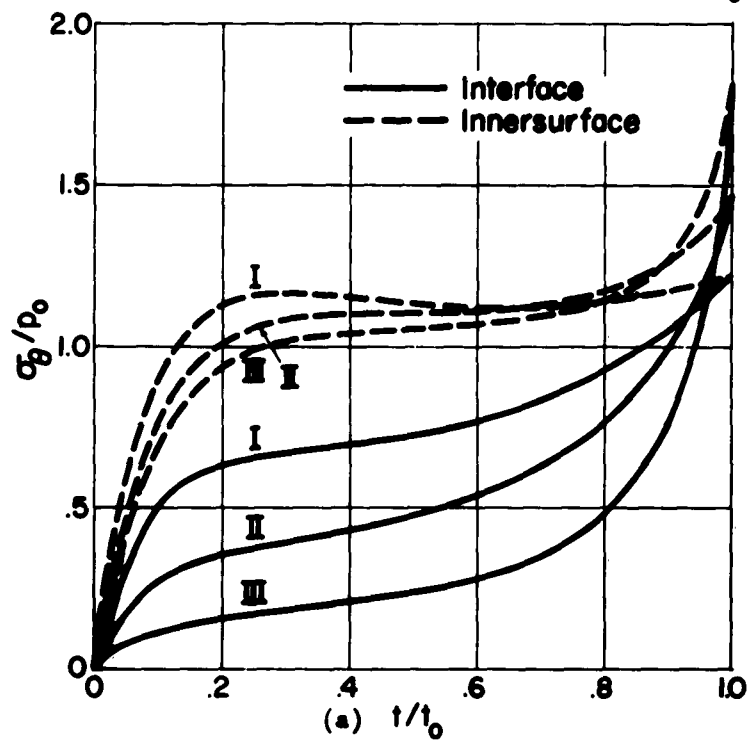


Figure 7. The Tangential Stress σ_θ at the Inner Surface and the Interface as a Function of Time t/t_0 .

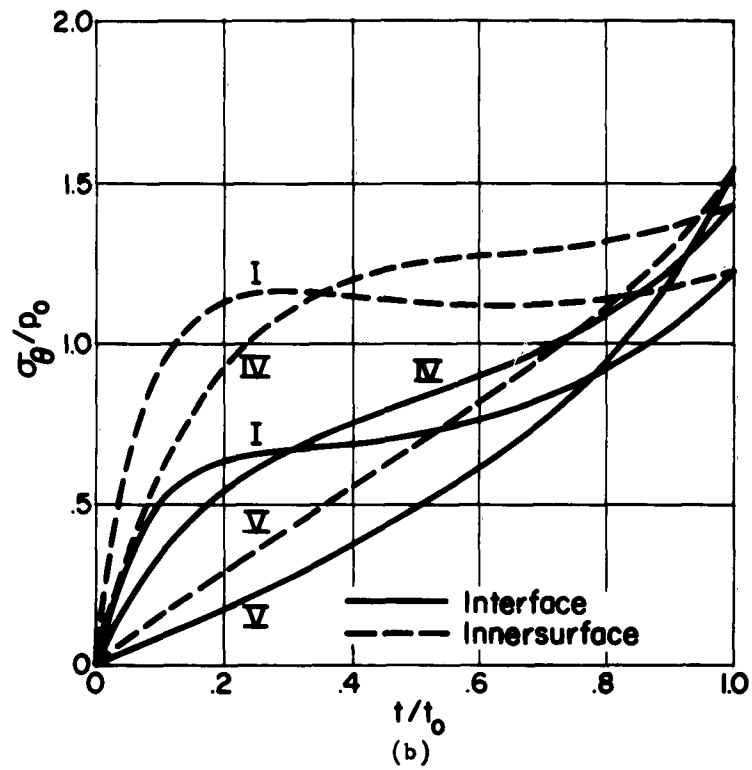
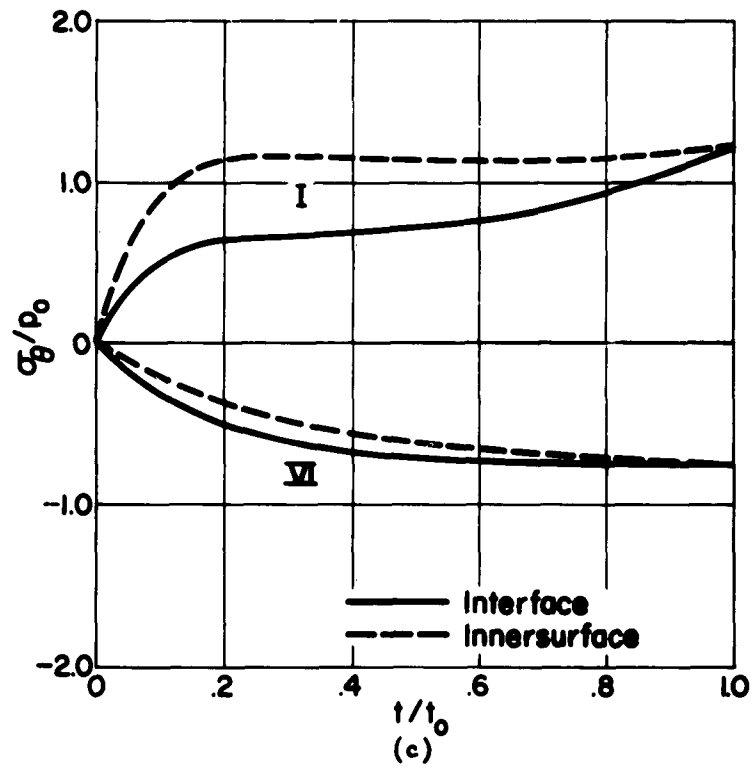


Figure 7. The Tangential Stress σ_θ at the Inner Surface and the Interface as a Function of Time t/t_0



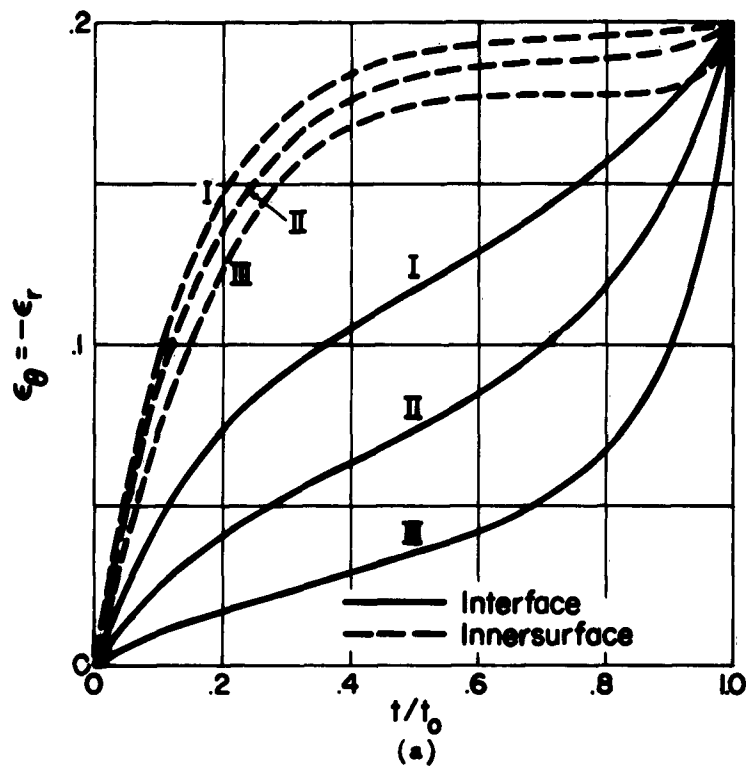
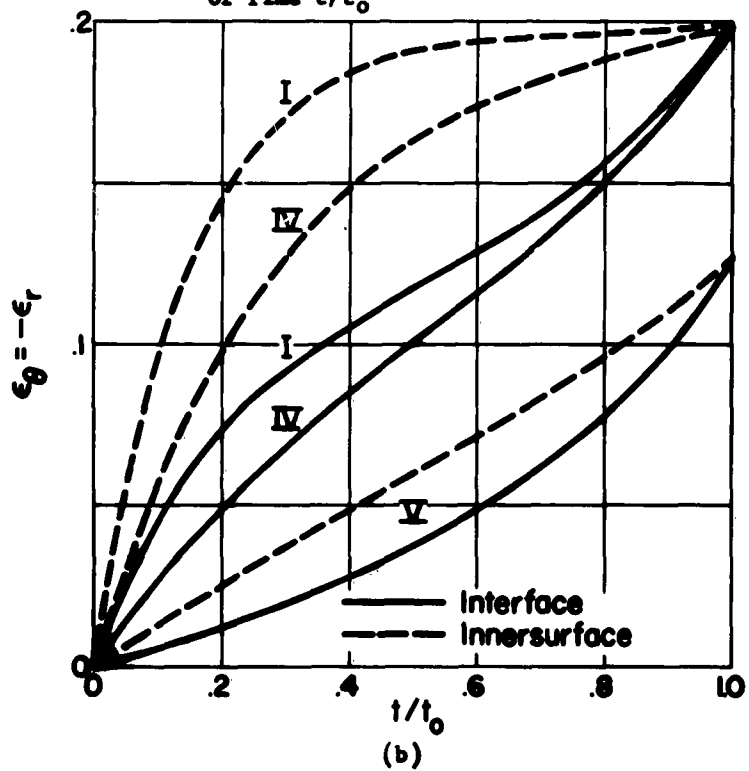


Figure 8. The Radial and Tangential Strains, ϵ_r and ϵ_θ , at the Inner Surface and Interface as Functions of Time t/t_0



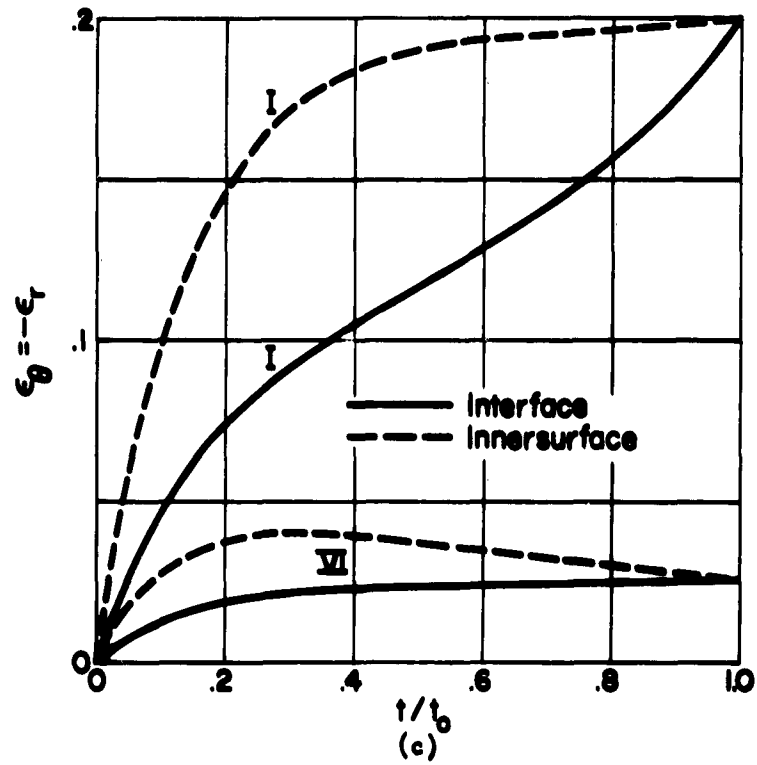


Figure 8. The Radial and Tangential Strains, ϵ_r and ϵ_θ , at the Inner Surface and the Interface as Functions of Time t/t_0

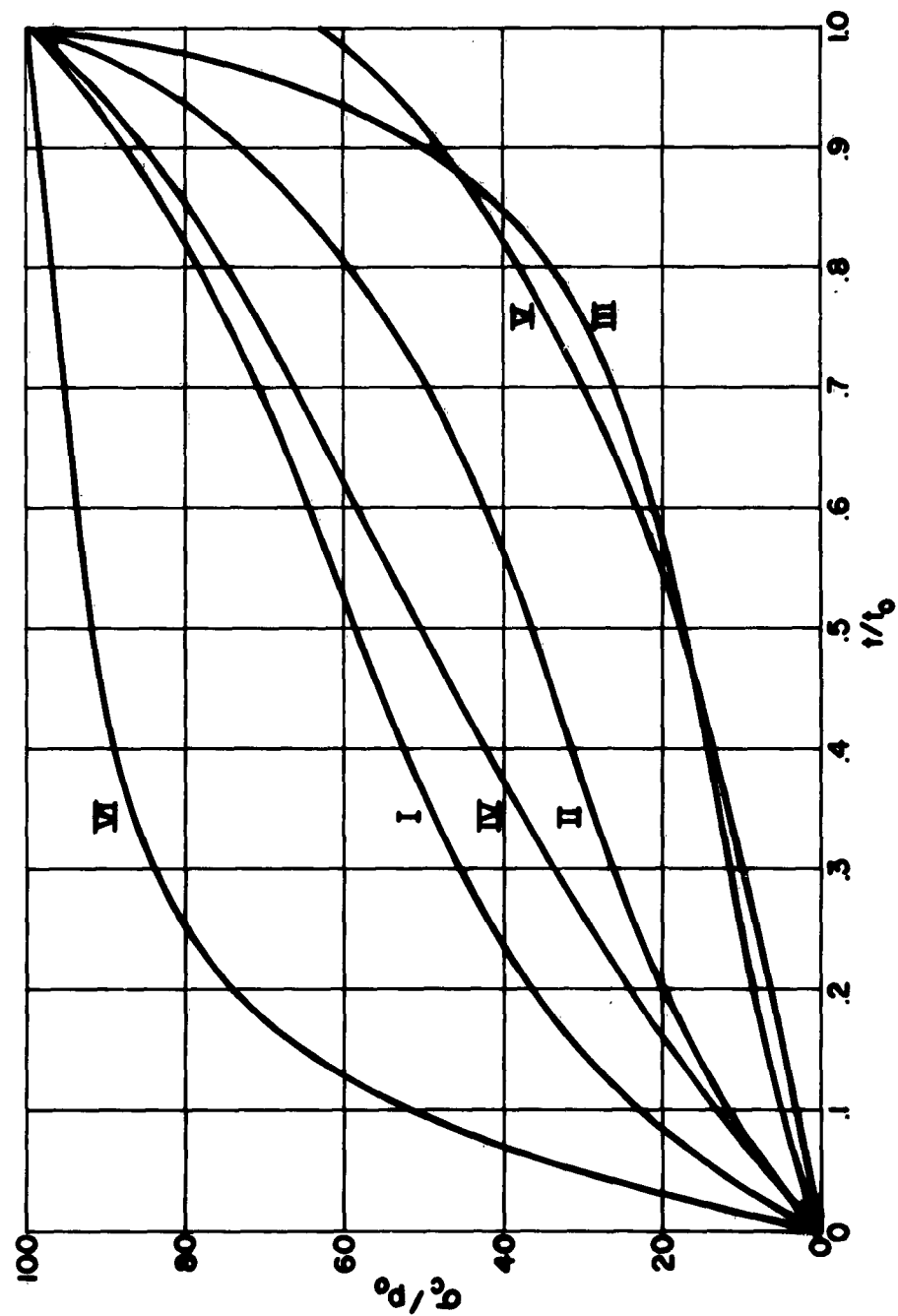


Figure 9. The Shell Stress σ_c as a Function of Time t/t_0

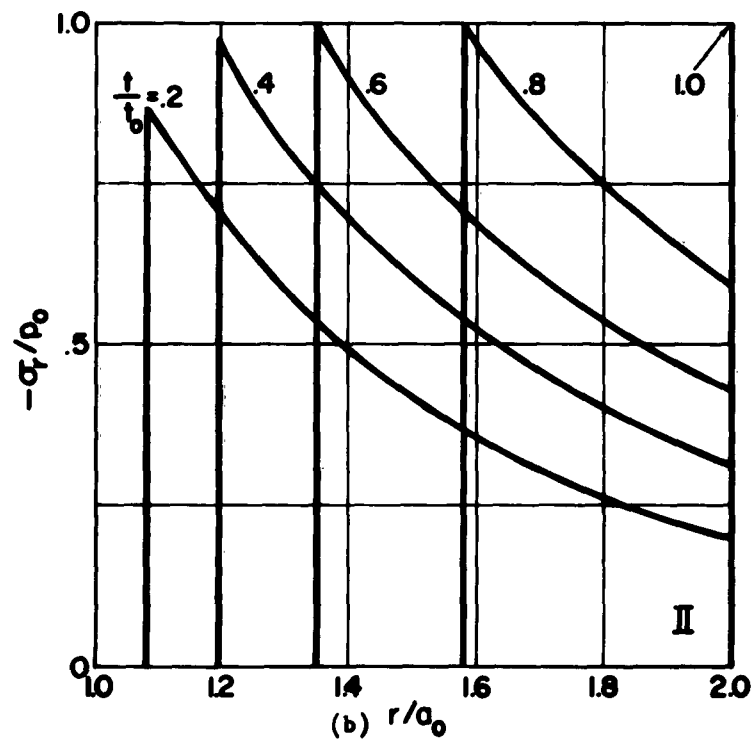
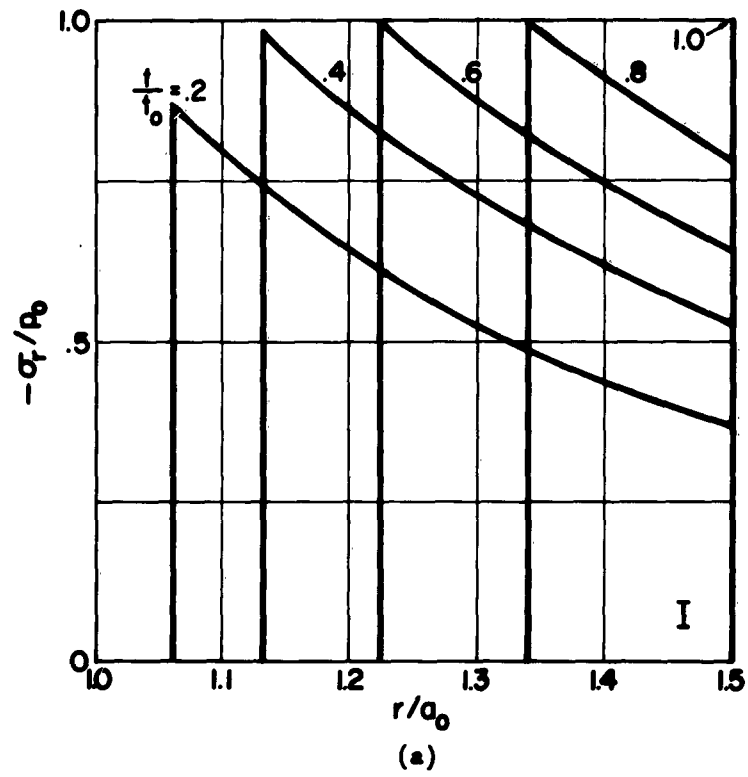


Figure 10. Space Distribution of Radial Stresses σ_r

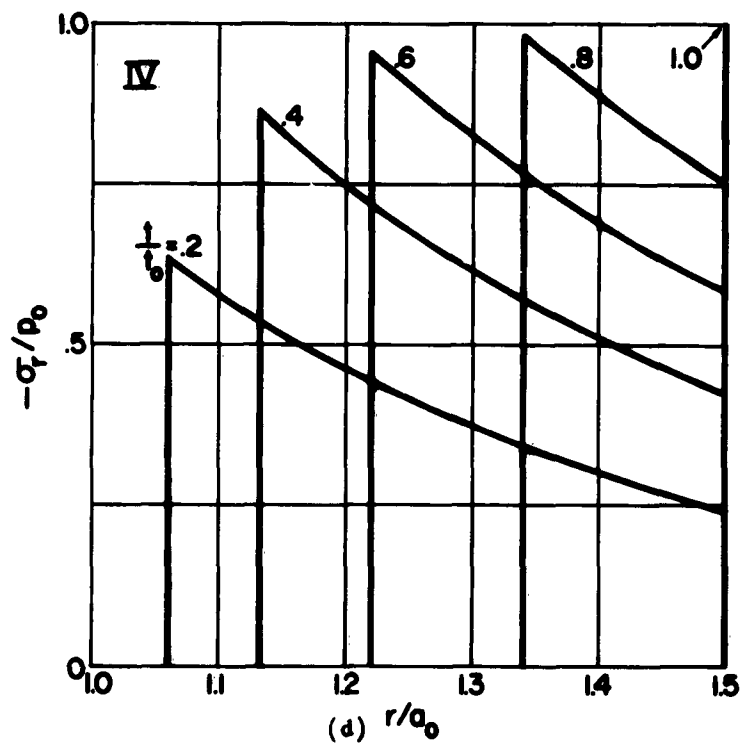
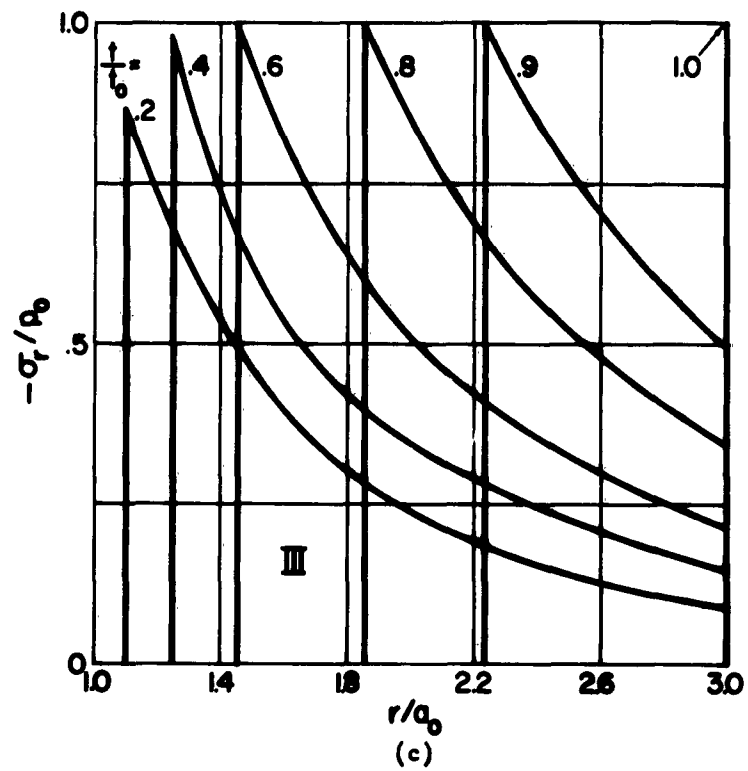


Figure 10. Space Distribution of Radial Stresses σ_r

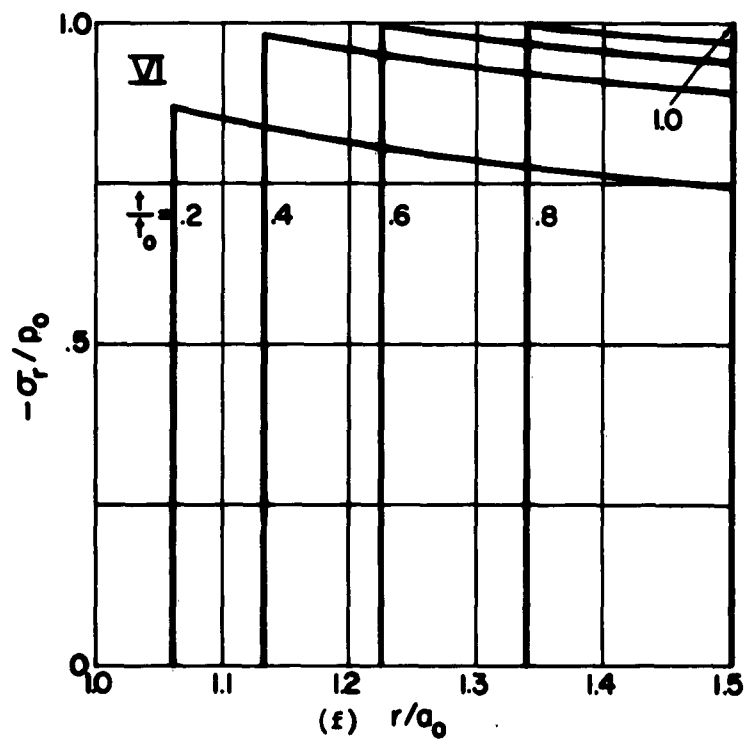
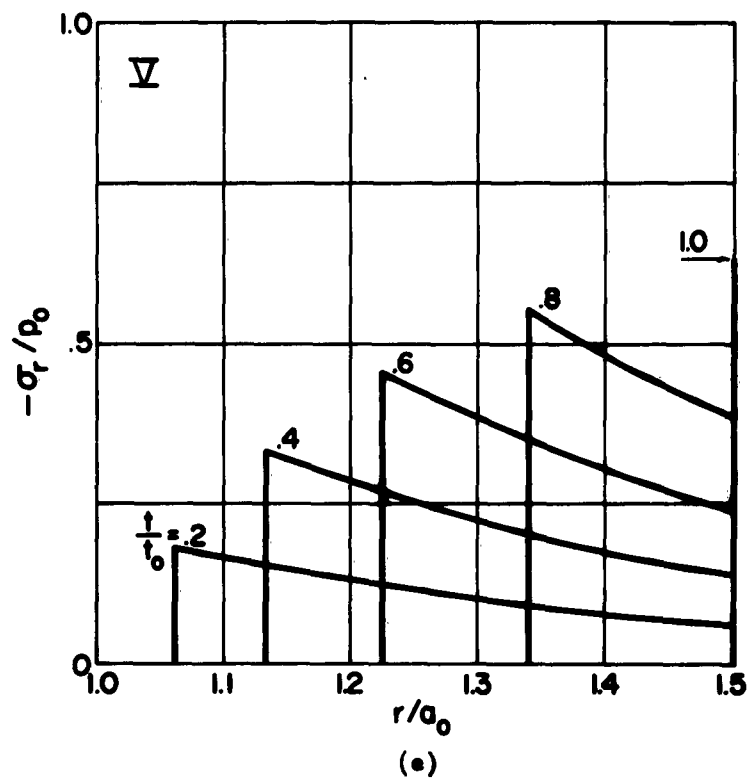


Figure 10. Space Distribution of Radial Stresses σ_r

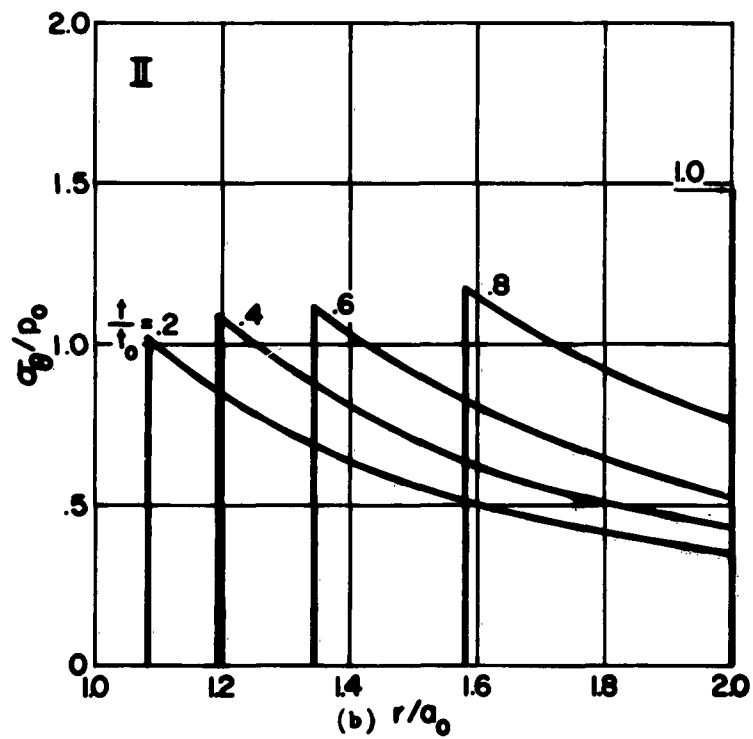
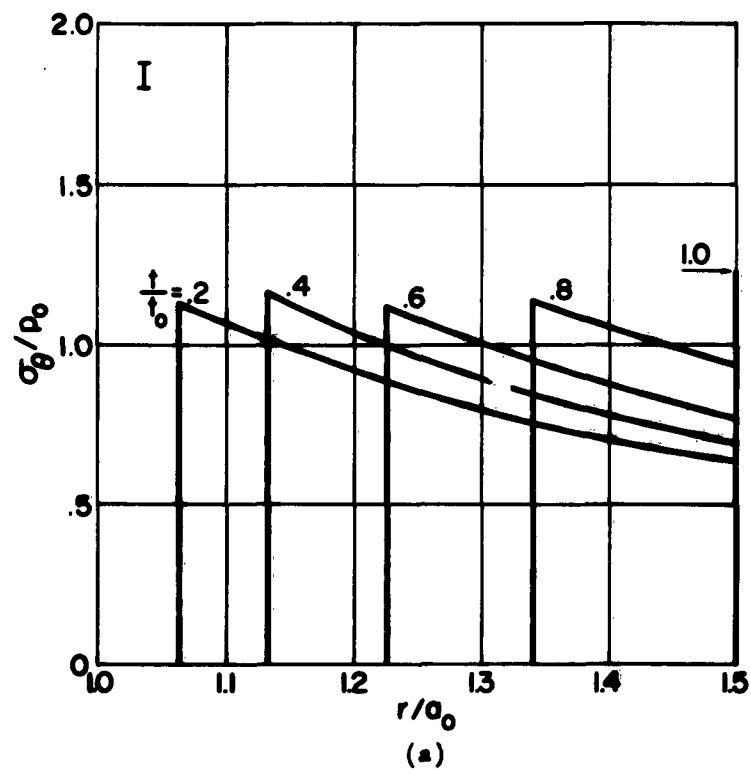


Figure 11. Space Distribution of Tangential Stresses σ_{θ}

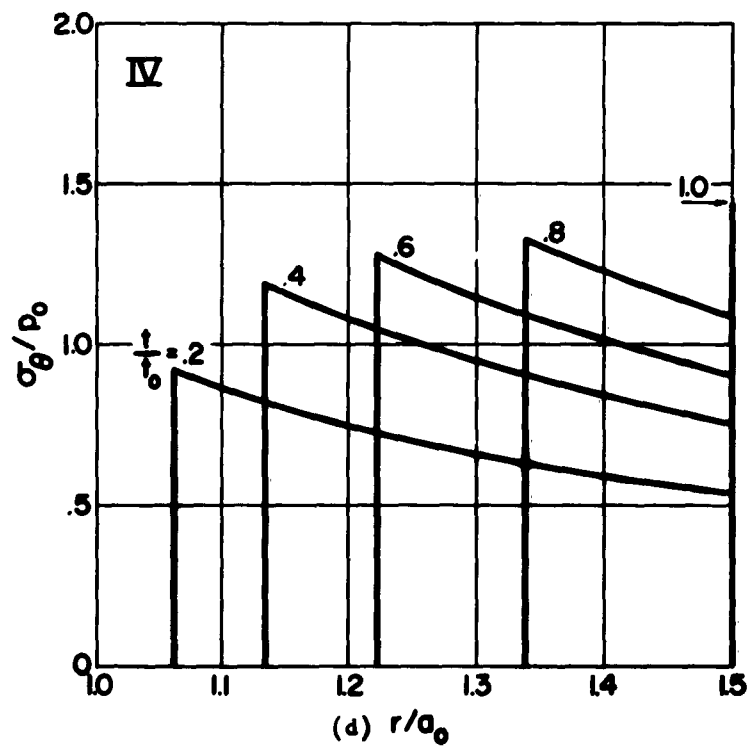
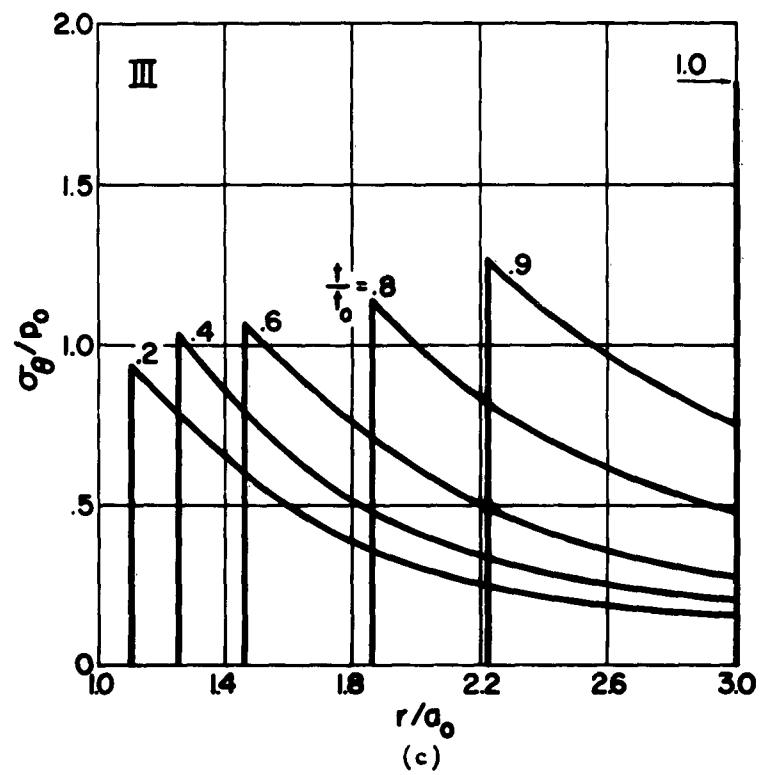


Figure 11. Space Distribution of Tangential Stresses σ_θ

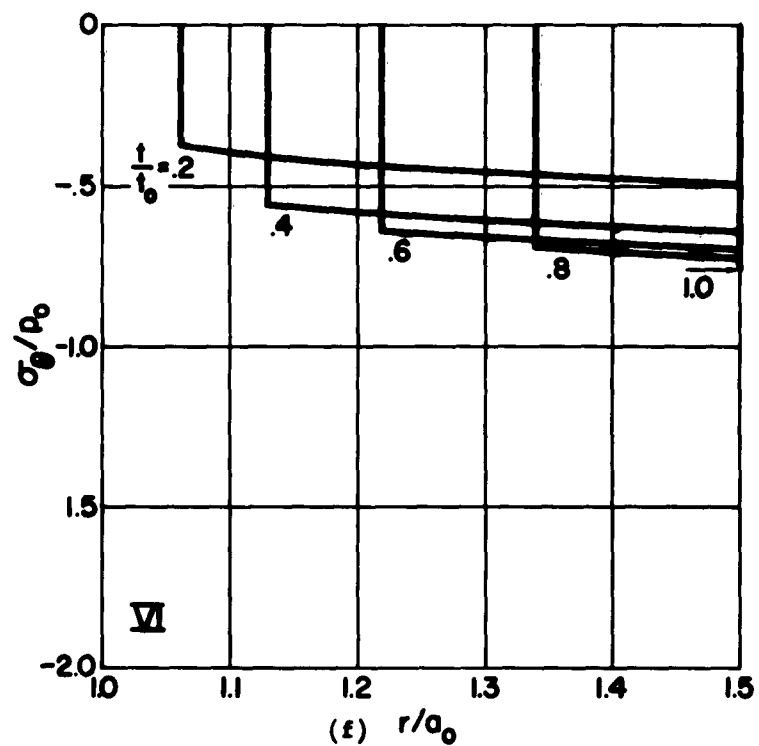
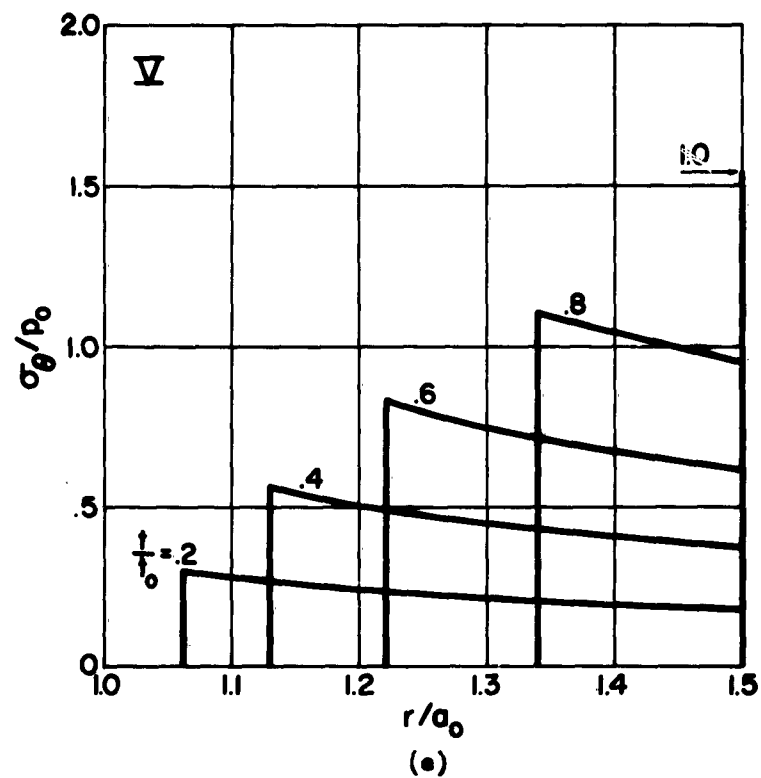


Figure 11. Space Distribution of Tangential Stresses σ_{θ}

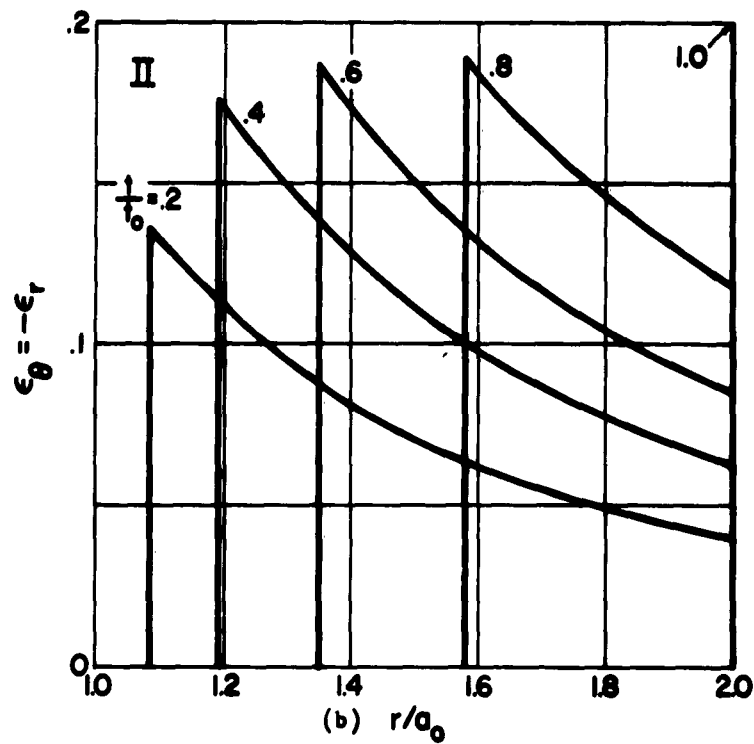
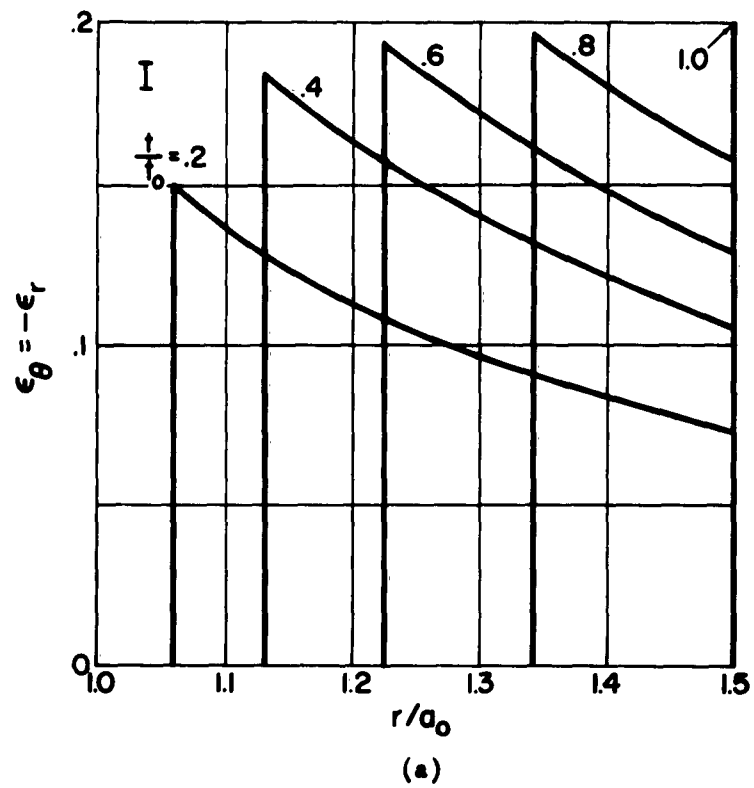


Figure 12. Space Distribution of Radial and Tangential Strains, ϵ_r and ϵ_θ

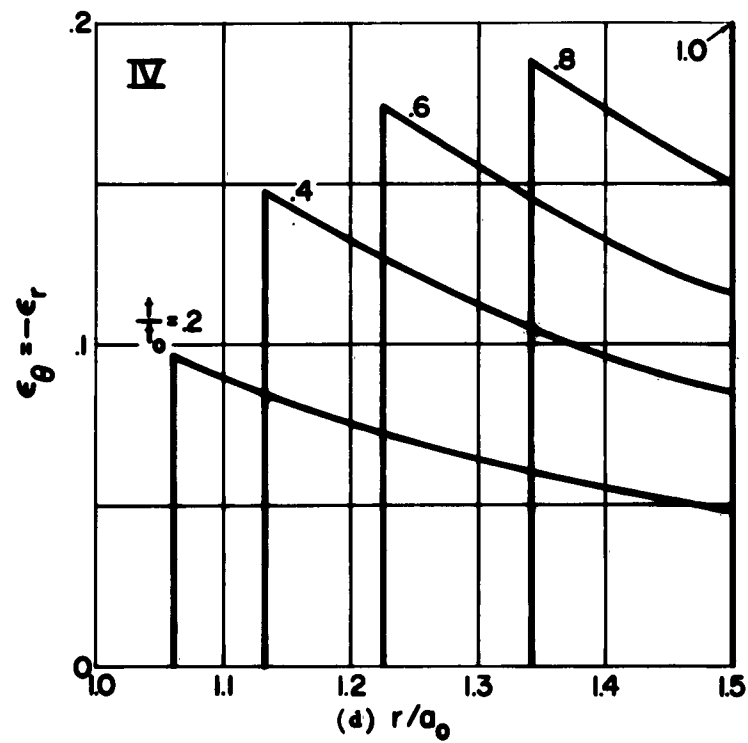
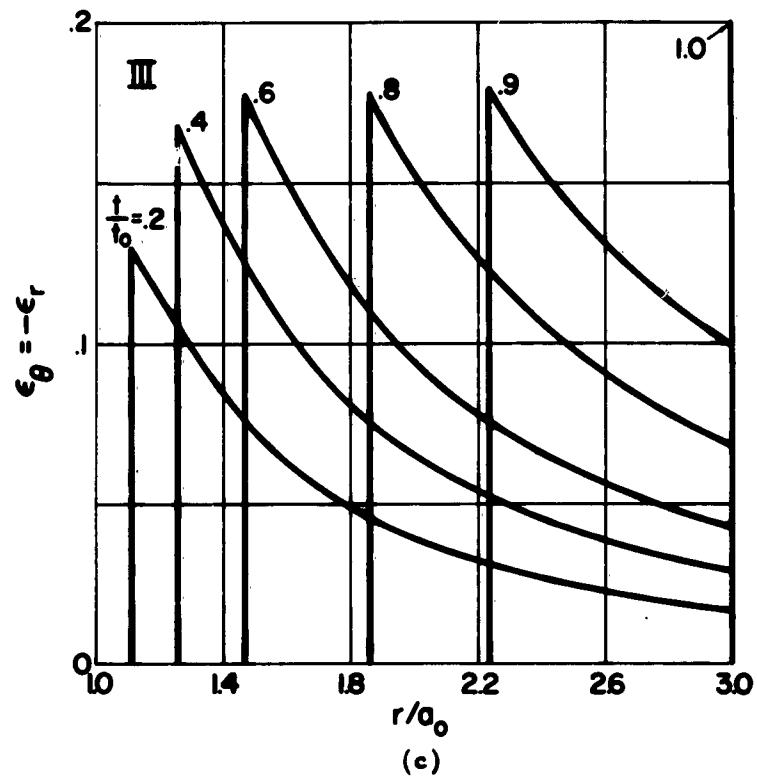


Figure 12. Space Distribution of Radial and Tangential Strains, ϵ_r and ϵ_{θ}

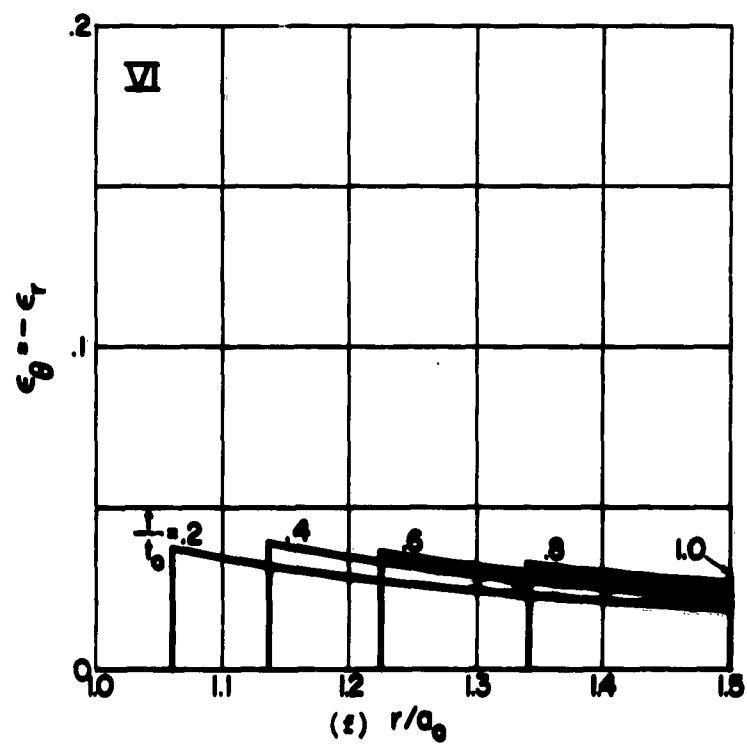
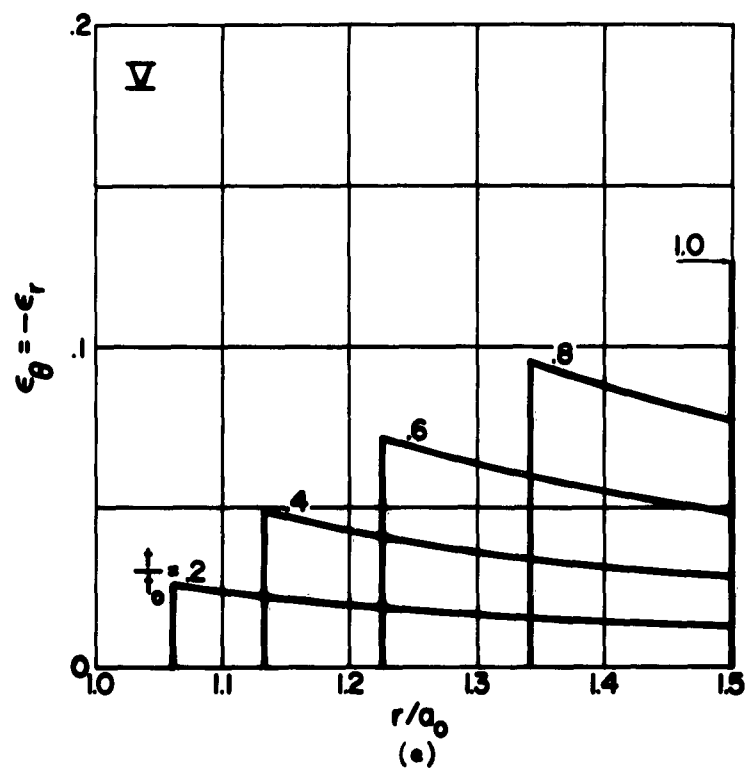


Figure 12. Space Distribution of Radial and Tangential Strains, ϵ_r and ϵ_{θ}

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